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# Transient Stability of Empty and Fluid-Filled Cylindrical Shells

In the present work a qualitatively accurate low dimensional model is used to study the non-linear dynamic behavior of shallow cylindrical shells under axial loading. The dynamic version of the Donnell non-linear shallow shell equations are discretized by the Galerkin method. The shell is considered to be initially at rest, in a position corresponding to a pre-buckling configuration. Then, a harmonic excitation is applied and conditions to escape from this configuration are sought. By defining steady state and transient stability boundaries, frequency regimes of instability may be identified such that they may be avoided in design. Initially a steady state analysis is performed; resonance response curves in the forcing plane are presented and the main instabilities are identified. Finally, the global transient response of the system is investigated in order to quantify the degree of safety of the shell in the presence of small perturbations. Since the initial conditions, or even the shell parameters, may vary widely, and indeed are often unknown, attention is given to all possible transient motions. As parameters are varied, transient basins of attraction can undergo quantitative and qualitative changes; hence a stability analysis which only considers the steady-state and neglects this global transient behavior, may be seriously non-conservative.

**Keywords**: Cylindrical shells, fluid-structure interaction, parametric instability, nonlinear vibrations

#### Introduction

Thin-walled cylindrical shells are widely used in many industries. Due to the increasing use of high-strength materials, sophisticated numerical techniques and optimization methods in analysis, the design of such shells is often buckling-critical. In many circumstances these shells are subjected not only to static loads but also to dynamic disturbances and filled with internal fluid. However, thin-walled cylindrical shells when subjected to axial compressive loads often exhibit a highly nonlinear behavior with a high imperfection sensitivity and may loose stability at loads levels as low as a fraction of the material strength.

Many studies are concerned with the analysis of shells vibrating in vacuum; far fewer are focused on the analysis of the nonlinear vibrations of cylindrical shells in contact with a dense fluid. One of the first studies on vibrations of circular cylindrical shells in contact with a dense fluid considering shell nonlinearity is due to Ramachandran (1979). He studied the large-amplitude vibrations of circular cylindrical shells having circumferentially varying thickness and immersed in a quiescent, non-viscous and incompressible fluid, using the Donnell's shell theory.

Boyarshina (1984, 1988) studied theoretically the nonlinear free and forced vibrations and stability of a circular cylindrical tank partially filled with a liquid and having a free surface. Here, nonlinearity is attributed to the interaction of free surface waves and elastic flexural vibrations of the shell.

Gonçalves and Batista (1988) considered simply supported circular cylindrical shells filled with incompressible fluid. To model the shell, Sanders' nonlinear shell theory and a novel mode expansion that includes two terms in the radial direction (the asymmetric and the axisymmetric ones) and ten terms to describe the in-plane displacements were used. Numerical results were obtained concerning the effect of the liquid on the nonlinear behavior of shells. It was found that the presence of a dense fluid increases the softening characteristics of the frequency-amplitude

relation when compared with the results for the same shell in vacuum.

Chiba (1993) studied experimentally large-amplitude vibrations of two vertical cantilevered circular cylindrical shells made of polyester sheets partially filled with water to different levels. He observed that for bulging modes with the same axial wave number, the weakest degree of softening nonlinearity can be attributed to the mode having the minimum natural frequency, as observed for the same empty shells. He also found that shorter tanks have a larger softening nonlinearity than taller ones, as in vacuum. The tank with a lower liquid height has a stronger softening nonlinearity than the same tank with a higher liquid level. Traveling wave modes and coupling between two bulging modes (and between two sloshing modes) were also observed.

Amabili et al (1998) studied the nonlinear free and forced vibrations of a simply supported, circular cylindrical shell in contact with an incompressible and non-viscous, quiescent dense fluid. Donnell's nonlinear shallow-shell theory is used, so that moderately large vibrations can be analyzed. The boundary conditions on radial displacement and continuity of circumferential displacement are exactly satisfied, while the axial constraint is satisfied "on the average". The problem is reduced to a system of ordinary differential equations by means of the Galerkin method, assuming an appropriate deflection shape. The mode shape is expanded by using two asymmetric modes (driven and companion modes) plus the axisymmetric mode.

In the present study, a low dimensional model which retains the essential nonlinear terms is used to study the nonlinear oscillations and instabilities of the shell. Here the interest is focused on a pivotal interaction between non-symmetric and axisymmetric modes which allows the escape from the pre-buckling configuration. To discretize the shell, Donnell shallow shell equations, modified with the transverse inertia force, are used together with Galerkin method to derive a set of coupled ordinary differential equations of motion. These equations are integrated numerically using the fourth order Runge-Kutta method. In order to study the nonlinear behavior of the shell, several numerical strategies were used to obtain time responses, Poincaré maps and bifurcation diagrams. The interested reader will find a description of the relevant numerical algorithms in Del Prado (2001).

It is considered a non-viscous and incompressible fluid and an irrotational flow. As a result, it can be characterized by a velocity

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potencial. The solution for the velocity potencial is taken as a sum of suitable functions, where the unknown parameters are determined by the kinetic condition along the shell wetted surface (Batista and Goncalves, 1988).

Steady state and transient stability boundaries are presented and special attention is devoted to the determination of the critical load conditions. From this theoretical analysis, dynamic buckling criteria can be property established which may constitute a consistent and rational basis for design of these shell structures under harmonic loading.

#### **Problem Formulation**

## **Shell Equations**

Consider a perfect thin-walled fluid-filled circular cylindrical shell of radius R, length L, and thickness h. The shell is assumed to be made of an elastic, homogeneous, and isotropic material with Young's modulus E, Poisson ratio V, and mass per unit area M. The axial, circumferential and radial co-ordinates are denoted by, respectively, x, y and z, and the corresponding displacements on the shell surface are in turn denoted by U, V, and W, as shown in Fig. 1.

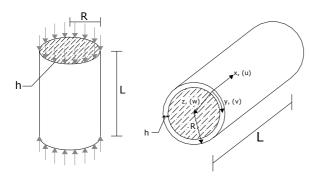


Figure 1. Shell geometry and coordinate system.

The shell is subjected to a uniformly distributed axial load given by:

$$P(t) = P_0 + P_1 \cos(\omega t) \tag{1}$$

where  $P_0$  is the uniform static load applied along the edges x=0, L,  $P_1$  is the magnitude of the harmonic load, t is time and  $\omega$  is the forcing frequency.

The nonlinear equations of motion based on the Von Karmán-Donnell shallow shell theory, in terms of a stress function f and the transversal displacement w are given by:

$$M\ddot{w} + \beta_{1}\dot{w} + \beta_{2}\nabla^{4}w = p_{h}R + F_{,yy}w_{,xx}$$

$$+ F_{,xx}\left(w_{,yy} + \frac{1}{R}\right) - 2F_{,xy}w_{,xy}$$
(2)

$$\frac{1}{Eh}\nabla^4 f = -\frac{1}{R}w_{,xx} - w_{,xx}w_{,yy} + w_{,xy}^2$$
 (3)

where

$$F = f^F + f$$
  $f^F = -\frac{1}{2}P_0 y^2$ 

and  $p_h$  is the fluid pressure,  $\nabla^4$  is the biharmonic operator,  $\beta_1$  and  $\beta_2$  are damping coefficients and D is the flexural rigidity defined as:

$$D = \frac{E h^3}{12 \left(1 - v^2\right)} \tag{4}$$

In the foregoing, the following non-dimensional parameters will be introduced:

$$W = \frac{w}{h} \qquad \qquad \xi = \frac{x}{L} \qquad \qquad \theta = \frac{y}{R}$$

$$\tau = \omega_o t \quad \Omega = \frac{\omega}{\omega_o} \quad \bar{f} = \frac{R}{E h^2 L^2} f$$

$$\Gamma_o = \frac{P_0}{P_-} = \frac{R\sqrt{3(1 - v^2)}}{E h^2} P_o \qquad \qquad \Gamma_1 = \frac{P_1}{P_-} = \frac{R\sqrt{3(1 - v^2)}}{E h^2} P_1$$
(5)

where  $\omega_0$  is the lowest natural frequency of the empty shell.

#### **Modal Analysis**

The numerical model is developed by expanding the transversal displacement component w in series in the circumferential and axial variables. From previous investigations on modal solutions for the non-linear analysis of cylindrical shells under axial loads (Hunt et al. 1986; Gonçalves and Batista, 1988; Gonçalves and Del Prado, 2002) it is observed that, in order to obtain a consistent modeling with a limited number of modes, the sum of shape functions for the displacements must express the non-linear coupling between the modes and describe consistently the unstable post-buckling response of the shell as well as the correct frequency-amplitude relation.

The dimensionless lateral deflection W can be generally described as (Gonçalves and Batista, 1988):

$$W = \sum_{i=1,3,5} \sum_{j=1,3,5} W_{ij} \cos(in\theta) \sin(jm\pi\xi) +$$

$$\sum_{k=0,2,4} \sum_{j=0,2,4} W_{ij} \cos(kn\theta) \sin(lm\pi\xi)$$
(6)

where n is the number of waves in the circumferential direction of the basic buckling or vibration mode, m is the number of half-waves in the axial direction,  $\theta = y/R$  and  $\xi = x/L$ .

These modes represent both the symmetric and asymmetric components of the shell deflection pattern. The first double series represents the unsymmetrical modes with odd multiples of the basic wave numbers m and n. The second double series represents, besides the asymmetric modes which contains an even multiple of the basic wave numbers m and n and rosette modes, the axysimmetric modes which play an important role in the non-linear modal coupling and loss of stability of the shell.

Previous studies on buckling of cylindrical shells have shown that the most important modes are the basic buckling or vibration mode and the axi-symmetric mode with twice the number of half waves in the axial direction as the basic mode, that is:

$$W = \zeta(\tau)_{11} \cos(n\theta) \operatorname{sen}(m\pi\xi) + \zeta(\tau)_{02} \cos(2m\pi\xi)$$
(7)

The relevance of these modes from a physical point of view was explained by Croll and Batista (1981) and, from symmetry and catastrophe theory arguments, by Hunt et al. (1986). These modes are enough to describe the initial post-buckling behavior of the shell as well as the topology of the pre-buckling configuration and the

potential barrier connected with the unstable equilibrium positions lying on the initial post-buckling path.

Substituting the assumed form of the lateral deflection, Eq. (7), on the right-hand side of the compatibility Eq. (3), this equation can be solved to obtain the stress function f in terms of w together with the relevant boundary and continuity conditions. Upon substituting the modal expressions for f and w into Eq. (2) and applying the Galerkin method, a set of non-linear ordinary differential equations is obtained in terms of modal amplitudes  $\zeta(\tau)_{ii}$ .

#### Fluid Mechanics Equations

The shell is assumed to be completely fluid-filled. The irrotational motion of an incompressible and non-viscous fluid can be described by a velocity potential  $\phi(x, r, \theta, t)$ . This potential function must satisfy the Laplace equation which can be written in dimensionless form as:

$$\overline{\phi}_{,\xi\xi} + \frac{1}{\kappa} \overline{\phi}_{,\kappa} + \frac{1}{\kappa^2} \overline{\phi}_{,\theta\theta} + \overline{\phi}_{,\kappa\kappa} = 0$$
 (8)

where

$$\kappa = r/R$$
  $\overline{\phi} = \gamma \phi/R^2$   $\gamma = [\rho_s R^2 (1 - v^2)/E]^{1/2}$ 

The dynamic fluid pressure acting on the shell surface is obtained from the Bernoulli equation:

$$p_{h} = -\left(\frac{\rho_{F}}{\rho_{S}}\right)\left(\frac{\gamma}{4\delta^{2}}\right)\overline{\phi}_{,t} \tag{9}$$

where  $\rho_F$  is the density of the fluid and  $\rho_S$  is the shell material density.

At the shell-fluid interface, the fluid velocity normal to the shell surface must be equal to the shell velocity in this direction, that is:

$$\overline{\phi}_{r} = 2\gamma \delta(\partial w/\partial t) \tag{10}$$

where  $\delta = h/2R$ .

Further, for a fluid-filled shell, the following restriction must be imposed at  $\kappa = 0$ :

$$\overline{\phi}_{x} = 0 \tag{11}$$

Solving equations (8) to (11), one obtains the following expressions for the hidrodynamic fluid pressure:

$$p_{h} = \zeta_{11,\sigma\tau} \, m_{\sigma} \cos\left(n\,\theta\right) \sin\left(m\,\pi\,\xi\right) \tag{12}$$

where  $m_a$  is the added mass due to the fluid contained in the shell, which is given by:

$$m_a = \left(\rho_F R\right) \left\{ m\pi \xi \left[ \frac{I_{n-1} \left(m\pi \xi\right)}{I_n \left(m\pi \xi\right)} - \frac{n}{m\pi \xi} \right] \right\}^{-1}$$
 (13)

where  $I_{n-1}$  and  $I_n$  are Bessel functions.

## Results

To check the validity and accuracy of the present methodology for the determination of the natural frequencies, a key point in any non-linear dynamic analysis, empty and fluid-filled cylindrical shells are analyzed and the results are compared with experimental and other numerical values found in literature. As a first example, the lowest natural frequencies of a simply supported empty cylinder are compared with the analytical solution derived by Dym (1973) using Sanders' shell theory and the experimental results obtained by Gasser (1987). The results are shown in Table 1. For the same shell, the present results for a water filled shell are compared with those obtained experimentally by Gasser (1987) and the numerical results obtained by Gonçalves and Batista (1987) in Table 2. In both cases, there is an excellent agreement between all results.

Table 1. Comparison of natural frequencies (Hz) for an empty cylindrical shell.(m = 1, L = 0.41 m, R = 0.3015 m, h = 0.001 m, E =  $2.1x10^8$  kN/m², v = 0.3,  $\rho = 7850$  kg/m³).

n	Gasser (1987)	Dym (1973)	Present work
7	318	305.32	303.35
8	278	281.37	280.94
9	290	288.28	288.71
10	334	317.51	318.40
11	362	362.22	363.33
12	418	417.96	419.19
13	478	482.23	483.51
14	550	553.67	554.97

Table 2. Comparison of natural frequencies (Hz) for a cylinder filled with water. (m = 1, L = 0.41 m, R = 0.3015 m, h = 0.001 m, E =  $2.1x10^8$  kN/m²,  $\nu = 0.3$ ,  $\rho = 7850$  kg/m³,  $\rho_F = 1000$  kg/m³).

n	Gasser (1987)	Gonçalves and Batista (1987)	Present work
8	120	118	119.2
9	124	124	127.9
10	146	144	146.7
11	182	171	173.3
12	214	204	206.4
13	254	243	245.0

Consider a thin-walled cylindrical shell with h=0.002m, R=0.2 m, L=0.4 m, E=2.1x10 $^8kN/m^2$ ,  $\nu$ =0.3,  $\beta_I$ =2 $\varepsilon M \omega_0$ , with  $\varepsilon$ =0.003 (fluid-filled shell) and  $\varepsilon$ =0.0008 (empty shell) (Popov et~al. 1998), and  $\beta_2$ = $\eta D$  with  $\eta$ =0.0001. The shell and fluid densities are:  $\rho_s$ =7850 $kg/m^3$  and  $\rho_F$ =1000 $kg/m^3$ . For this shell geometry the lowest natural frequency occurs for (n,m)=(5, 1).

Now the parametric instability and escape from the pre-buckling configuration of the fluid-filled cylinder under axial harmonic forcing, as described by Eq. (1), will be considered. In the following, the constant part of the loading ( $\Gamma_0$ ) is assumed to be between the upper and lower static critical load of the shell. In these circumstances, the shell potential energy exhibits three wells, one associated with the fundamental pre-buckling configuration and two wells associated with the two possible post-buckling configurations. If the cylinder is subjected to a periodic axial load, it will undergo the familiar longitudinal forced vibration, exhibiting a uniform transversal motion, due to the effect of Poisson's ratio, also known as breathing mode. However, at certain critical values, the longitudinal motion becomes unstable and the cylinder executes transverse bending vibrations.

Figure 2 shows some representative time histories for  $\Gamma_0$ =0.40. Here  $\Omega$ = $\omega/\omega_0$  and  $\omega_0$  is the lowest natural frequency of the unloaded shell. A projection of the phase space and Poincaré section are also shown in these figures. These figures were obtained by numerically integrating the equation of motion with the Runge-Kutta method. In Fig. 2.a, for a forcing amplitude lower than a critical value ( $\Gamma_1$ =0.45 and  $\Omega$ =1.00), after a finite initial disturbance, the amplitude of the response decreases rapidly converging to the trivial solution. If the

control parameter  $\Gamma_1$  is increased beyond a critical value, the shell exhibits initially an exponential growth of the amplitude, as shown in Fig. 2.b, converging to a limit cycle within the pre-buckling well. In this case, the trivial solution becomes unstable (parametric instability) and the system converges to a period-two stable solution. If  $\Gamma_1$  is increased to a higher value, for example  $\Gamma_1$ =1.30, the shell escapes from the pre-buckling well (snap-through buckling) and exhibits large cross-well chaotic motions, as shown in Fig. 2.c, or small amplitude oscillation around a post-buckling configuration.

Figure 3 shows the numerically obtained parametric instability boundary as well as the transient and permanent escape boundaries for the fluid-filled shell and the same shell in vacuum, in (frequency of excitation x amplitude of excitation) control space for  $\Gamma_0$ =0.40,  $\Gamma_0$ =0.60 and  $\Gamma_0$ =0.80. The lower stability boundary corresponds to parameter values for which small perturbations from the trivial solution will result in an initial growth in the oscillations; therefore it defines the parametric instability boundary. The second limit corresponds to escape from the pre-buckling potential well in a slowly evolving environment. These curves were obtained by increasing slowly the amplitude while holding the frequency constant. As one can observe, the parametric stability boundary is composed of various "curves", each one associated with a particular bifurcation event. The deepest well is associated with the principal instability region at  $\omega$ =2 $\omega_p$ , while the second well to the left is the

secondary instability region occurring around  $\omega = \omega_p$  and the other smaller wells to the left are connected with super-harmonic resonances. The horizontal dotted line corresponds to the static critical load of this shell. Comparing Figures 3.a, 3.b and 3.c, one can conclude that the static pre-loading has the effect of lowering the stability boundaries, of enlarging the width of the instability regions and of shifting the instability regions to the left. In both cases the instability boundaries can be much lower than the static critical load. The fluid has a similar influence on the stability boundaries. This is expected since the influence of the fluid is to increase the effective mass of the system, decreasing consequently the natural frequencies.

For the region between the parametric instability limit and the transient escape limit, the shell exhibits vibrations in the prebuckling potential well during both permanent and transient states. When comparing the permanent and transient boundaries, one can observe that the transient escape limit is lower than the permanent one. This means that the shell may exhibit large amplitude vibrations during the transient state but converge to a low amplitude solution within the pre-buckling well when the steady-state response is reached. A structure may display in a nonlinear regime long transients, but their lengths can not be known a priori. So, in order to avoid any damage due to large amplitude vibrations the transient response of the shell must be analyzed in detail.

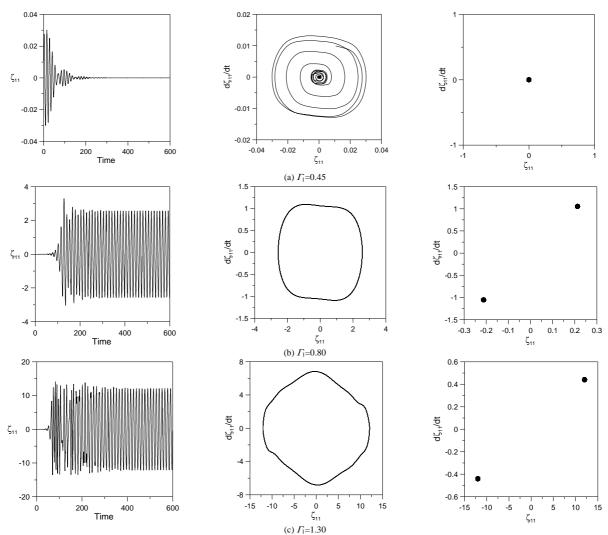


Figure 2. Time response, phase plane and Poincaré section for  $\Gamma_0$ =0.40 and  $\Omega$ =1.00. Fluid-filled circular cylindrical shell.

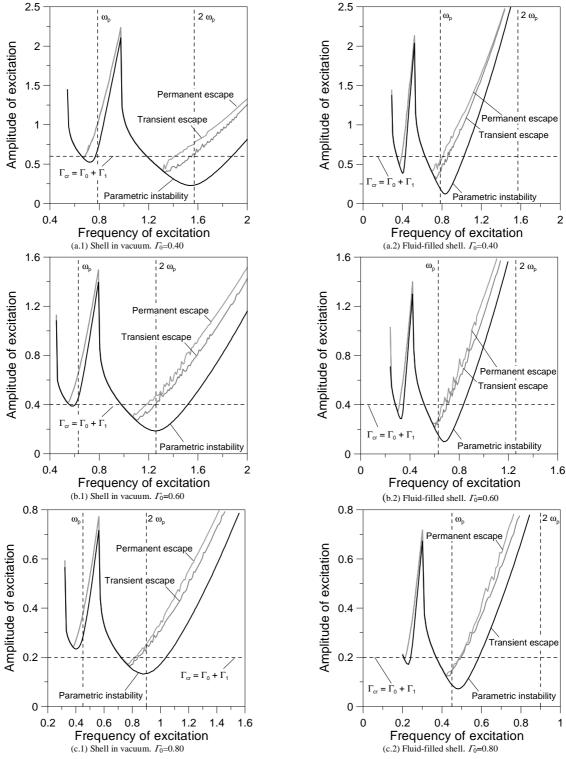


Figure 3. Instability boundaries in control space for different values of static load.

Figure 4 shows typical bifurcation diagrams connected with the principal instability region for the fluid-filled shell as a function of the forcing amplitude  $\Gamma_1$ , for different values of the forcing frequency  $\Omega$ . These bifurcation diagrams where obtained by numerical continuation techniques (Del Prado, 2001). In these diagrams a dotted line means unstable solutions and a continuous

line means stable solutions. The bifurcation diagram depicted in Fig. 4.a is typical of the left descending branch of the principal region of parametric instability. The system exhibits a sub-critical bifurcation, that is, the fundamental solution looses its stability, giving rise to a 2T unstable periodic motion. In this case, any increase in  $\Gamma_1$  beyond the critical value leads to a jump to another stable solution that may exist within the pre-buckling well or around a post-buckling

configuration. Also, the 2T solution exhibits a stable branch between two unstable branches. So, for load levels lower than the critical value the shell may display different types of behavior within the pre-buckling well. As observed in Fig. 4.a, this non-trivial stable region corresponds to forcing values lower than the critical load. This leaves a regime where there is no attractor within the pre-

buckling well after the critical point is reached and hence an unavoidable jump to escape under increasing forcing occurs. This explains why in this region the numerically obtained parametric instability boundary practically coincides with the transient and permanent escape boundaries.

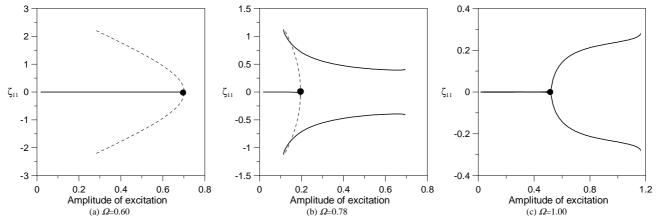


Figure 4. Bifurcation diagrams of the Poincaré map. Principal instability region for fluid-filled shell,  $\Gamma_0$ =0.40.

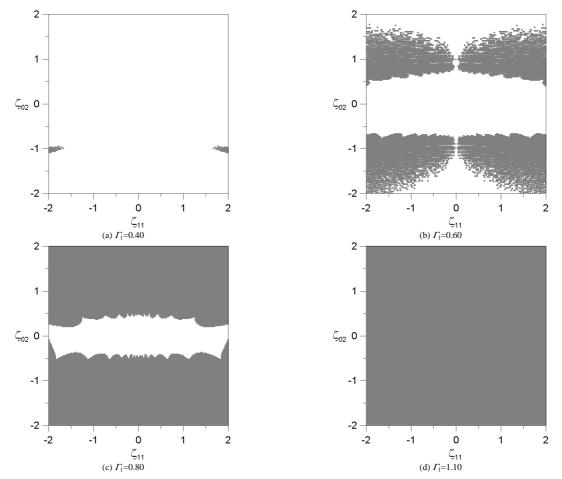


Figure 5. Cross sections of the basins of attraction, in transient state, for increasing values of the forcing amplitude  $\Gamma_1$  of the fluid-filled cylindrical shell. Evolution of the basin for  $\Gamma_0$ =0.40 and  $\Omega$ =1.00. Parametric instability load=0.525.

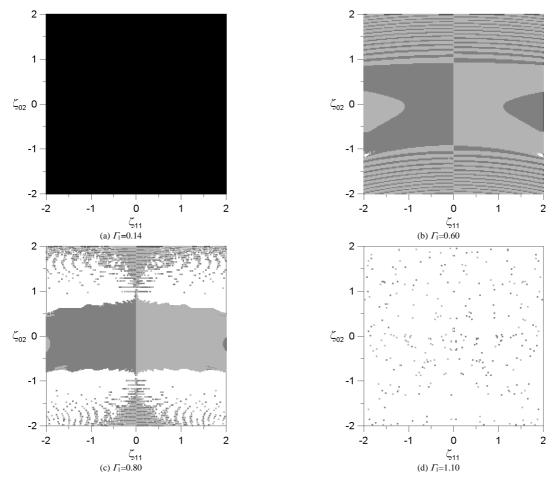


Figure 6. Cross sections of the basins of attraction, in permanent state, for increasing values of the forcing amplitude  $\Gamma_1$  of the fluid-filled cylindrical shell. Evolution of the basin for  $\Gamma_0$ =0.40 and  $\Omega$ =1.00. Parametric instability load=0.525.

In Figure 4.b, the jump is indeterminate. The bifurcation is subcritical, but the stable small-amplitude non-trivial solution subsists for forcing values higher than the critical load. So, when the fundamental trivial solution becomes unstable, the response may restabilize within the pre-buckling well or jump to a remote attractor. The response that is attained physically depends on the initial conditions. The bifurcation diagram shown in Fig. 4.c is typical of the right ascending branch of the stability boundary. When  $\Gamma_1$  is lower than the critical value, the only possible steady state solution within the pre-buckling well is the trivial one, which is stable. Consequently, the response is trivial. When  $\Gamma_1$  is greater than a critical value, there are two possible steady state solutions: (a) the trivial one, which is unstable; and (b) a finite amplitude period-two (2T) solution, which is stable. In this case, since disturbances are always present, the response is non-trivial. Also, these figures show that as  $\Gamma_1$  increases from zero, the response consists of the trivial solution. As  $\Gamma_1$  exceeds the critical value,  $\zeta_{11}$  begins to increase slowly with increasing  $\Gamma_1$ . The critical value in this case is a supercritical bifurcation. As the amplitude of the forcing increases, the amplitude of the response increases until the escape boundary is reached. Before escape occurs, the period-two solution may also become unstable, being followed by a period doubling cascade, eventually reaching a narrow zone of chaotic motion.

In order to evaluate the safety of the structure one should analyze the behavior of the basins of attraction of the solutions in both transient and permanent states. Figure 5 shows the evolution of the transient basin of attraction for increasing values of the forcing amplitude  $\Gamma_1$ ,  $\Omega$ =1.00 and  $\Gamma_0$ =0.40. Here the  $\zeta_{11}\times\zeta_{02}$  cross-sections of the four dimensional phase space  $(\dot{\zeta}_{11}=\dot{\zeta}_{02}=0.0)$  are shown for increasing values of the forcing amplitude. Figure 6 shows the evolution of the permanent basin of attraction for increasing values of the forcing amplitude  $\Gamma_1$ ,  $\Omega$ =1.00 and  $\Gamma_0$ =0.40. Both figures are associated with the bifurcation diagram of Fig. 4.c and cover the same set of initial conditions.

In Figure 5 the gray area is associated with the escape during the transient response and the white area corresponds to the fundamental trivial and period-two stable solutions within the prebuckling well. As  $\Gamma_1$  increases the region associated with the escape increases and after a certain critical value, it covers completely the analyzed region, showing that escape occurs for any set of initial conditions during the transient response, well before the critical escape load displayed in the bifurcation diagram of Fig. 4.c is reached.

In Figure 6 the black area corresponds to the fundamental trivial solution, the gray areas correspond to the period-two stable solution within the pre-buckling well and the white area corresponds to the escape. For  $\Gamma_1$  lower than the critical point, the response for initial conditions within the analyzed area converges to the trivial solution or to escape. Of course, escape can only occur for large perturbations. After the critical point, the black region suddenly disappears and the response for the majority of initial conditions converges to the period-two stable solutions within the pre-buckling well. As  $\Gamma_1$  increases, the region associated with this solution

decreases and a rapid erosion is observed. Also, after a certain critical value the whole basin of attraction becomes fractal. In this case the response becomes very sensitive to the initial conditions and the steady state response, unpredictable.

Comparing the trivial and period-two areas of Fig. 5 and Fig. 6, one can observe that the basin area occupied by the transient response is smaller than the area occupied by the permanent response. So, a practical design criterion must be based on the transient analysis rather than on the steady state response of the system. Also, the critical loads obtained from the bifurcation diagrams are not enough to evaluate the robustness of the structure in the presence of unavoidable disturbances occurring during its construction or service life. The analysis of size and structure of the basin of attraction must be taken into account in order to specify allowable disturbances in a dynamic environment. A detailed parametric analysis of the basin evolution considering empty and fluid-filled shells can be found in Silva (2004).

# **Concluding Remarks**

Based on Donnell's shallow shell equations, an accurate lowdimensional model is derived and applied to the study of the nonlinear vibrations of an axially loaded fluid-filled circular cylindrical shell in transient and permanent states. The results show the influence of the modal coupling on the post-buckling response and on the nonlinear dynamic behavior of fluid-filled circular cylindrical shells. Also the influence of a static compressive loading on the dynamic characteristics is investigated with emphasis on the parametric instability and escape from the pre-buckling configuration. The most dangerous region in parameter space is obtained and the triggering mechanisms associated with the stability boundaries are identified. Also the evolution of transient and permanent basin boundaries is analyzed in detail and their importance in evaluating the degree of safety of a structural system is discussed. It is shown that critical bifurcation loads and permanent basins do not offer enough information for design. Only a detailed analysis of the transient response can lead to safe lower bounds of escape (dynamic buckling) loads in the design of fluidfilled cylindrical shells under axial time- dependent loads.

## Acknowledgements

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